

SELF-POWERED WIRELESS SENSORS

Fred Dynys and Ali Sayir , NASA Glenn Research Center, USA

NASA's integrated vehicle health management (IVHM) program offers the potential to improve aeronautical safety, reduce cost and improve performance by utilizing networks of wireless sensors. Development of sensor systems for engine hot sections will provide real-time data for prognostics and health management of turbo-engines. Sustainable power to embedded wireless sensors is a key challenge for prolong operation. Harvesting energy from the environment has emerged as a viable technique for power generation.

Thermoelectric generators provide a direct conversion of heat energy to electrical energy. Micro-power sources derived from thermoelectric films are desired for applications in harsh thermal environments. Silicon based alloys are being explored for applications in high temperature environments containing oxygen. Chromium based p-type Si/Ge alloys exhibit Seebeck coefficients on the order of $160\ \mu\text{V/K}$ and low thermal conductance of 2.5 to 5 W/mK. Thermoelectric properties of bulk and thin film silicides will be discussed.

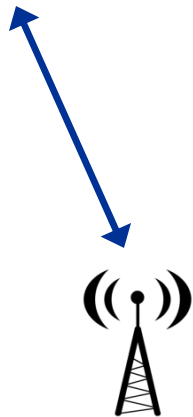


Self-Powered Wireless Sensors

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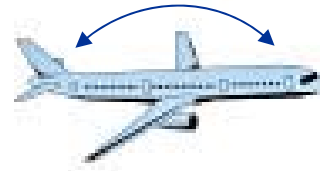
Integrated Vehicle Health Monitoring

Real Time Health
Diagnostics From
Remote Airplane



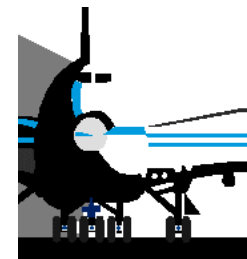
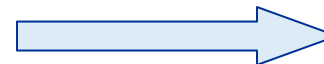
Airport

Examine Sensor Data for Early
Detection of Component Failure



Wireless Sensor
Network to Monitor
Vehicle System

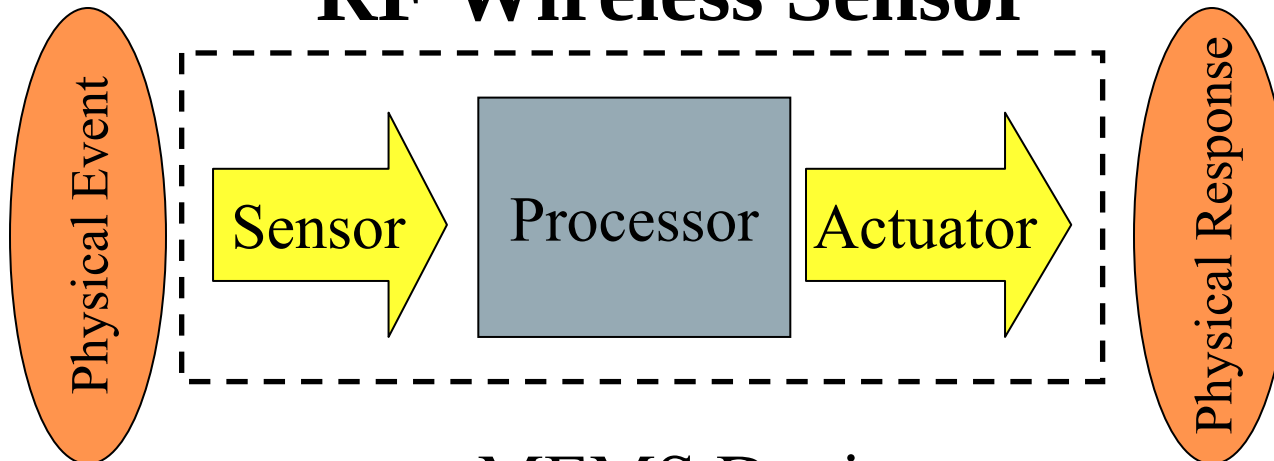
Repair



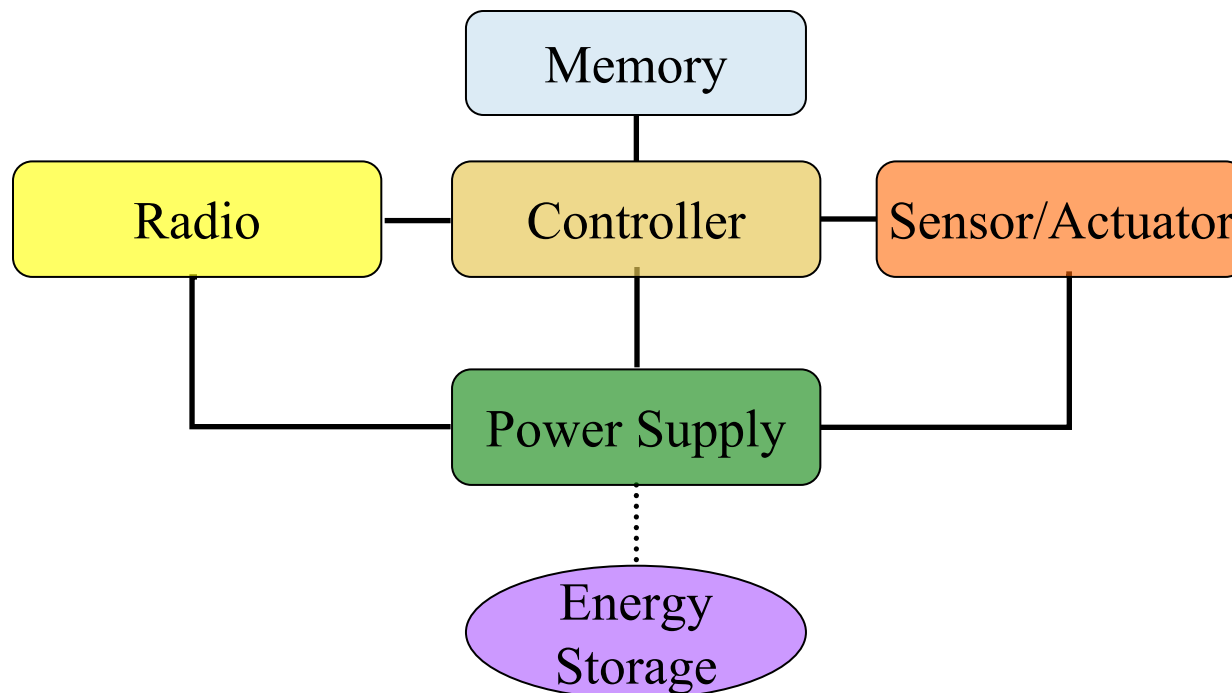
Proactive Maintenance

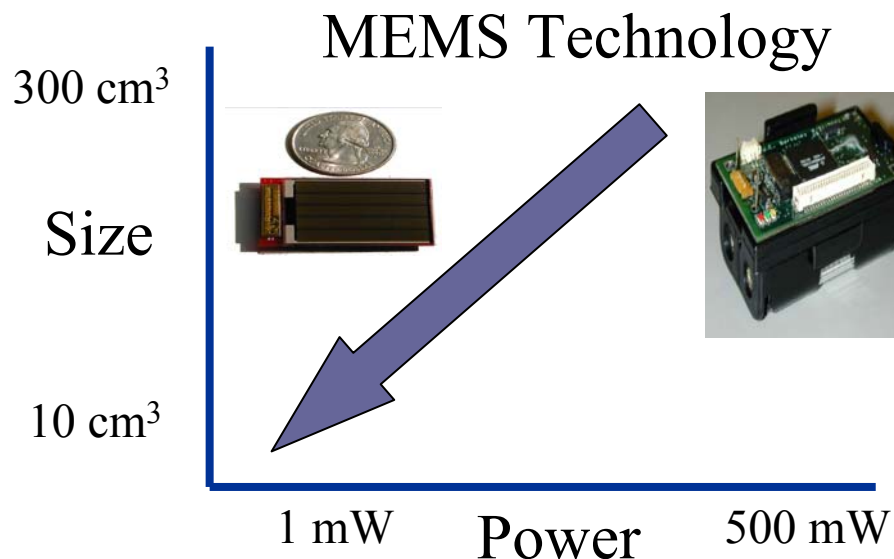
Future airplanes will have a wireless sensor network for monitoring vehicle system health, flight environment and structural integrity.

RF Wireless Sensor



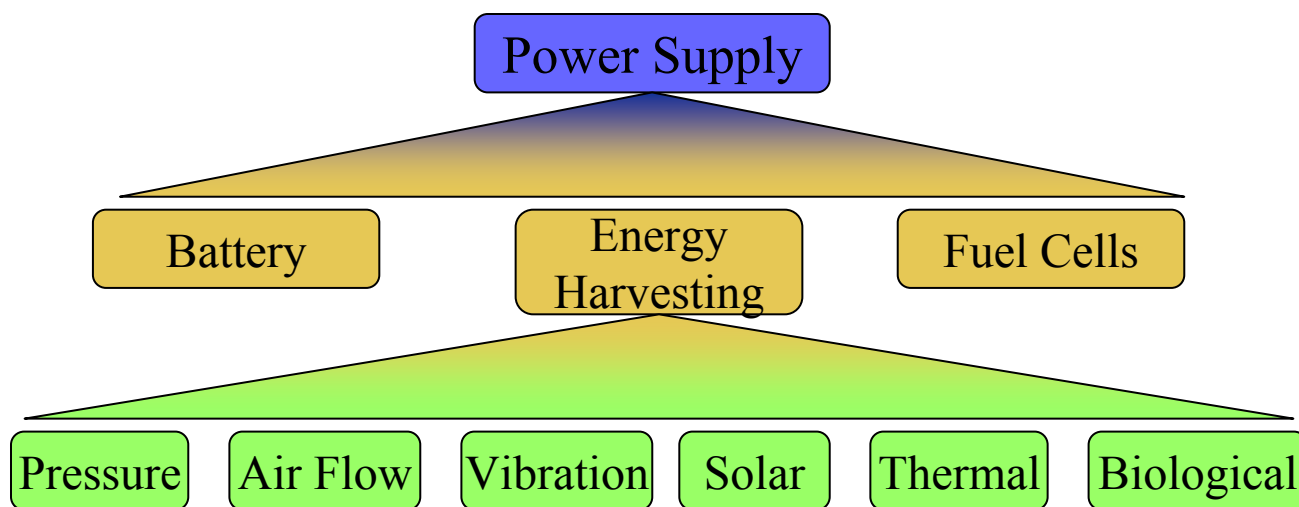
MEMS Device





Energy Harvesting

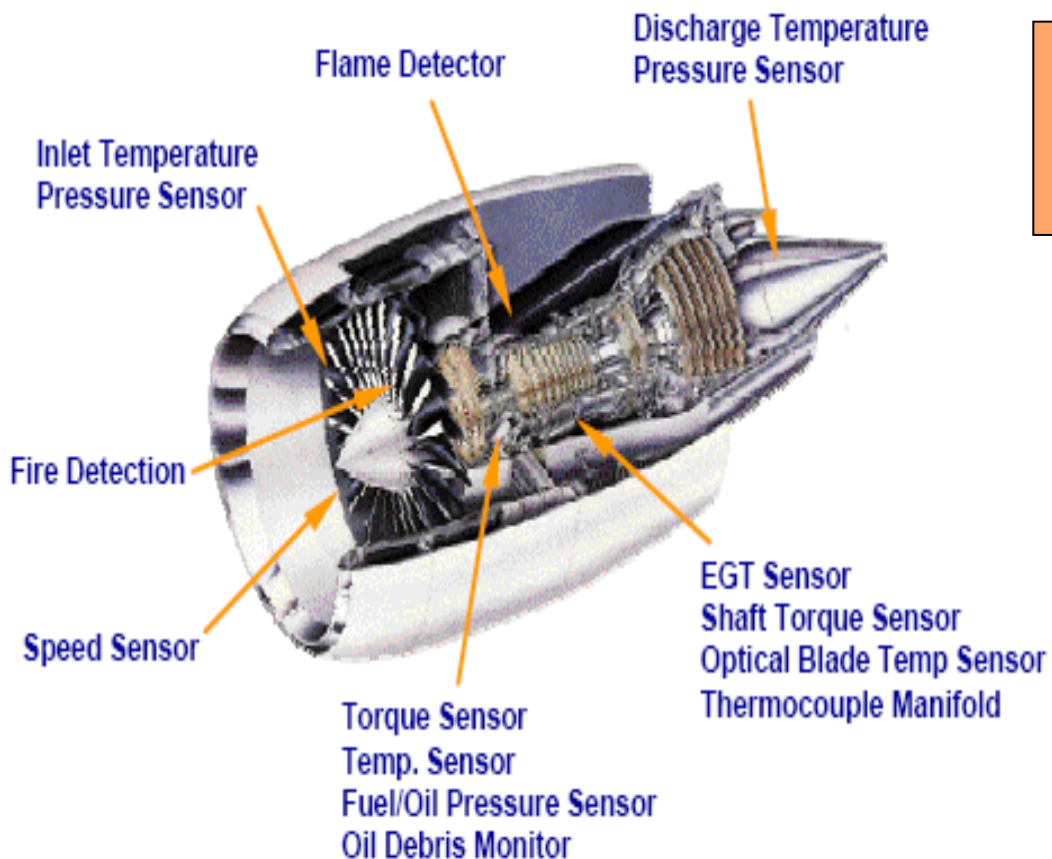
- Reduction in MEMS energy consumption is the key to enabling energy harvesting
- Looking for renewable energy sources in the application environment.
- Infinite power source.



	$\mu\text{W}/\text{cm}^3$
Solar	15 - 15000
Air Flow	~380
Biological	~330
Vibration	100 - 400
Thermal	<100
Pressure	<100

Wireless Sensors

Wireless technology allows sensors to be placed in remote locations



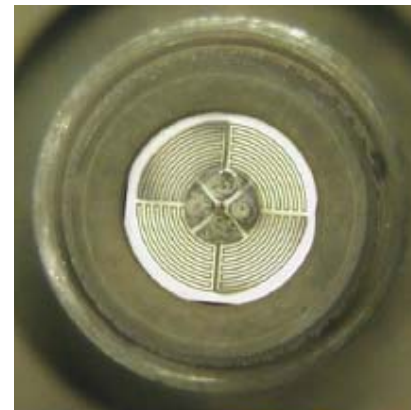
Sensors/power supply fabricated from materials that are operational in high temperature environments

GRC Sensors

Temperature



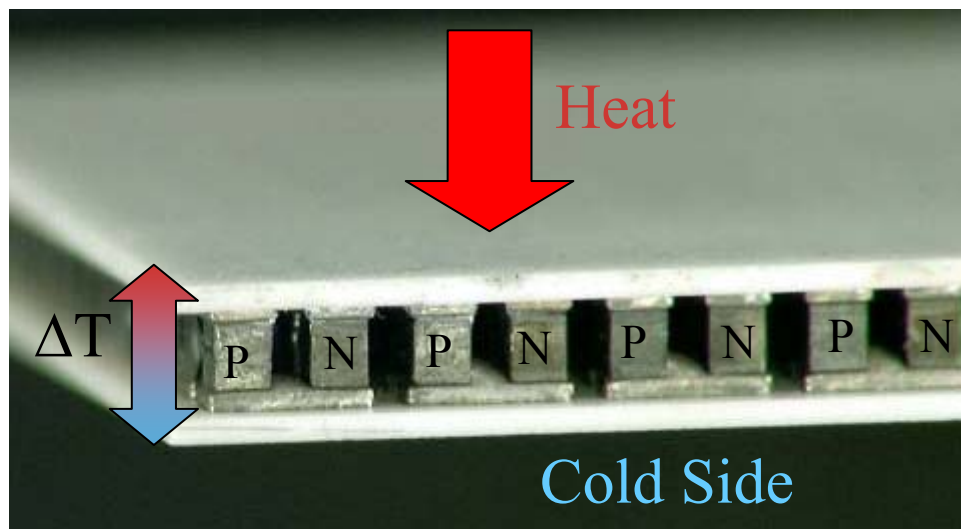
Heat Flux



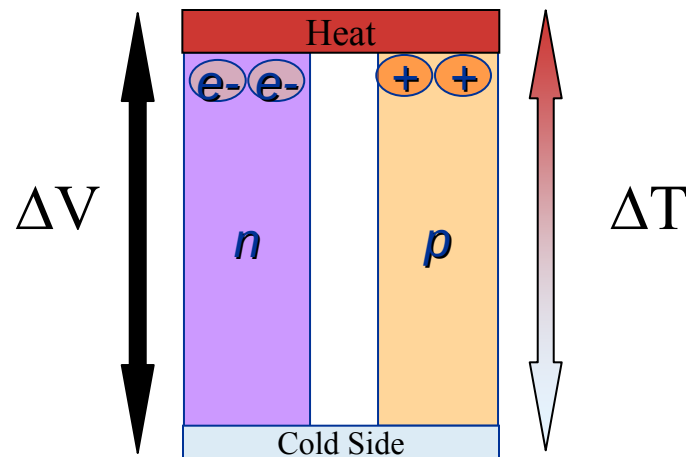
J. Wrbanek & G. Fralick

Self powered wireless sensors require technological advancement in energy harvesting

Thermoelectric Converters

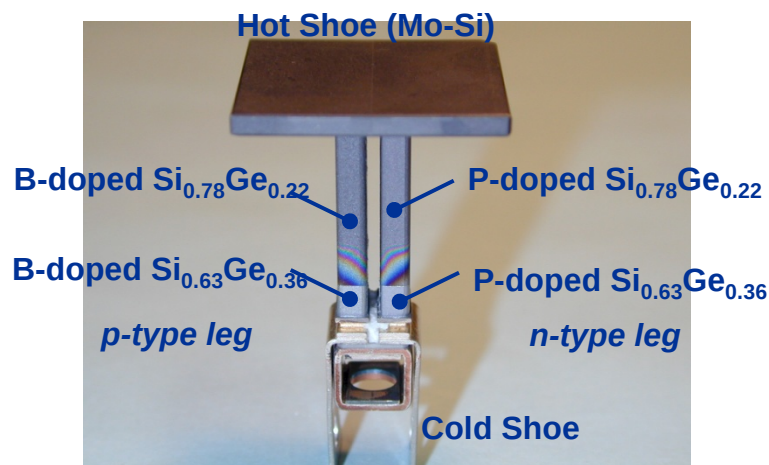


Seebeck Effect



Low Efficiency (<7%) – Niche Markets

Radioisotope Thermoelectric Generator



Proven Technology

- Long Life-Continuous Power- 20+ years
- Durable in Harsh Environments
- No Moving Parts

Thermoelectrics

Figure of Merit

$$ZT = \frac{S^2 \sigma}{\kappa} T$$

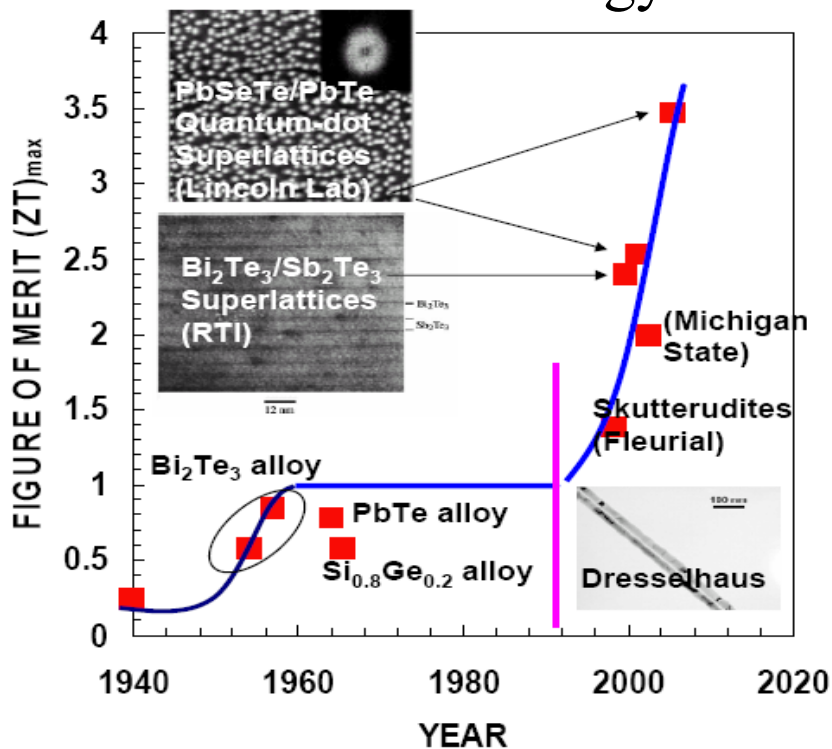
Desired Properties of TE's

- High Seebeck coefficient (S)
- High electrical conductivity (σ)
- Low thermal conductivity (κ)

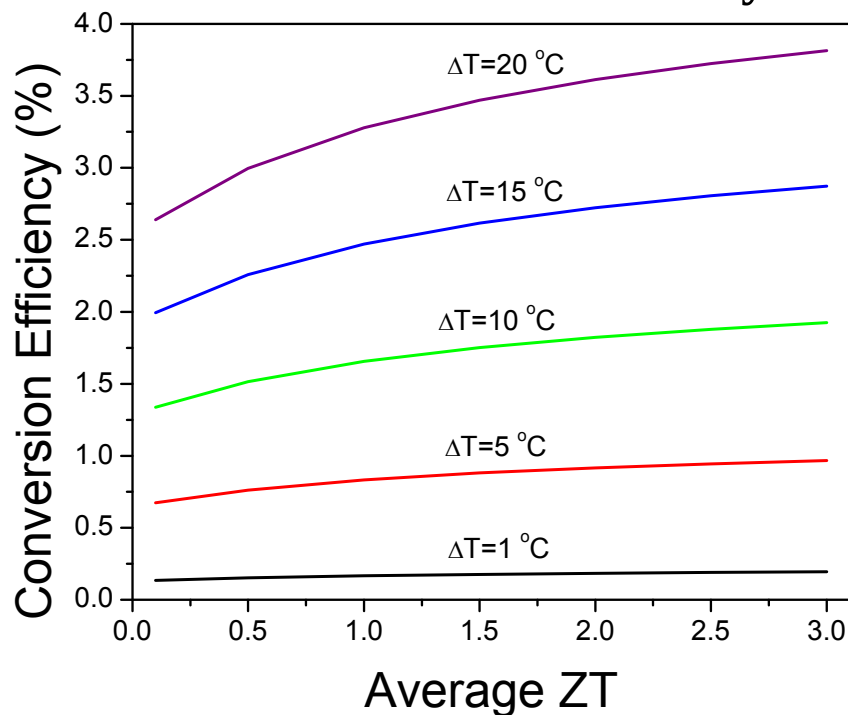
Efficiency

$$\eta_{\max} = \frac{\Delta T}{T_{\text{hot}}} \frac{\sqrt{1+ZT}-1}{\sqrt{1+ZT}+T_{\text{cold}}/T_{\text{hot}}}$$

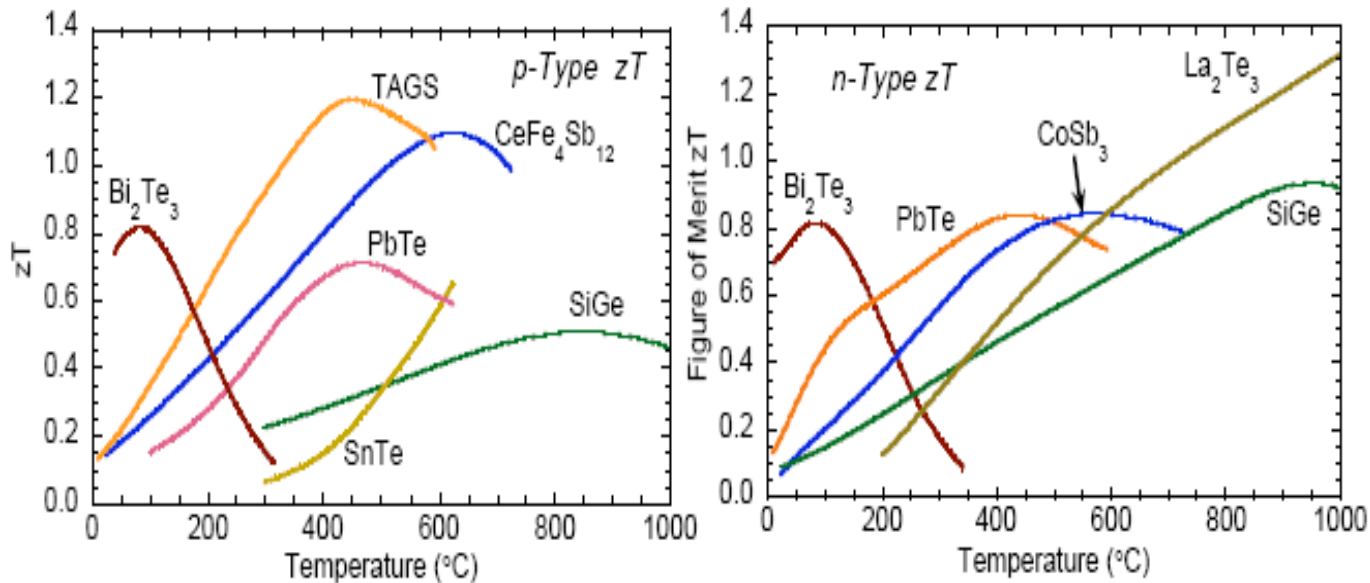
Nanotechnology



Low ΔT -Low Efficiency



Standard Thermoelectric Materials



Operational Conditions

- Temperatures 300 - 600 $^{\circ}\text{C}$
- Environment – Varying P_{O_2}



Silicides

- Environmental/Chemical Stability
- Low Electrical Resistance
- Low Formation Temperature



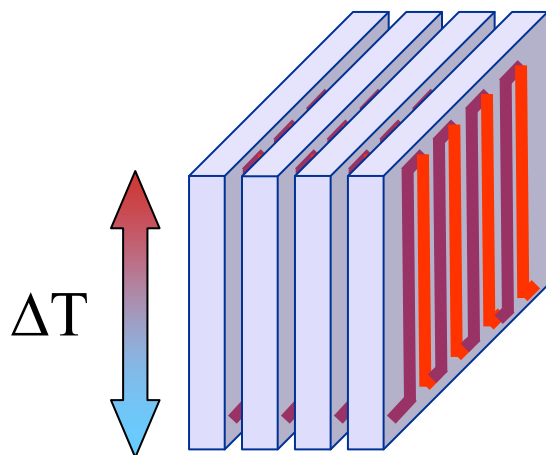
Electrical Transport

Semiconductor/Metallic Behavior

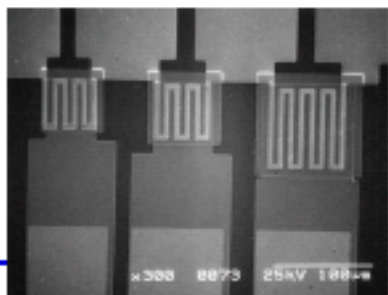
IIIA	IVA	VA	VIA	VIIA	VIII A			IB
	TiSi ₂	V ₃ Si V ₃ Si ₂ V ₅ Si ₃ VSi ₂	Cr ₃ Si Cr ₅ Si ₃ CrSi ₂	MnSi _{1.7}	α-FeSi ₂ β-FeSi ₂	Co ₂ Si CoSi CoSi ₂	Ni ₂ Si NiSi NiSi ₂	Cu ₃ Si
YSi ₂	ZrSi ₂	Nb ₃ Si NbSi ₂			Ru ₂ Si ₃	RhSi ₂	Pd ₂ Si	
	HfSi ₂	TaSi ₂		ReSi _{1.7}	OsSi ₂	IrSi IrSi ₃	PtSi	

	Band Gap	Type	Mobility Cm ² /SV	K W/mK @300 K	S mV/K @500K	ρ 10 ⁻⁵ Ω-m @500K	TE 10 ⁻⁶ /°C
CrSi ₂	~0.35 ev	P	15	6.8	220	7.2	11-14
MnSi _{1.7}	~0.7 ev	P	40	2.9	160	3.1	16
CoSi	-----	N	----	14.3	-50	0.2	11.1

Thermoelectric Module I



Compact geometry permits parallel connection of modules for reduced series resistance.



Micro-Cooler

TE Materials

- Thickness ~ 10 's μm
- Electrical conductivity $\geq 10^4 \Omega^{-1}\text{m}^{-1}$
- Larger ΔT
- Thick film stresses

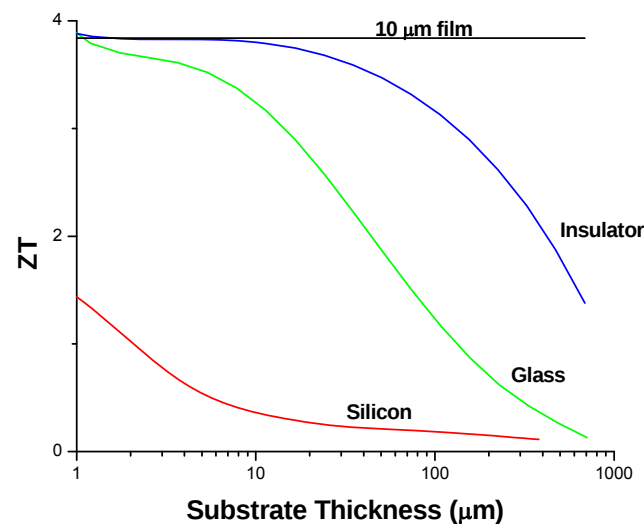
Substrate

- Thickness
- Low thermal conductivity
- Electrically insulating

Interconnect Technology

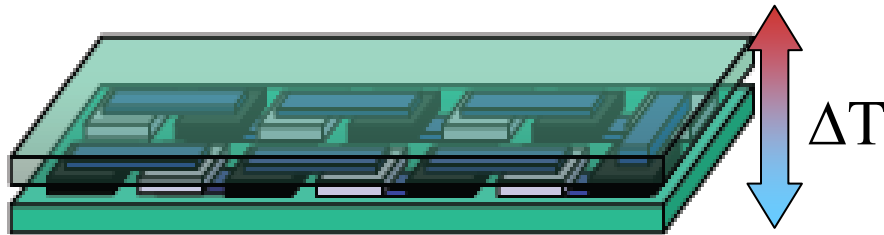
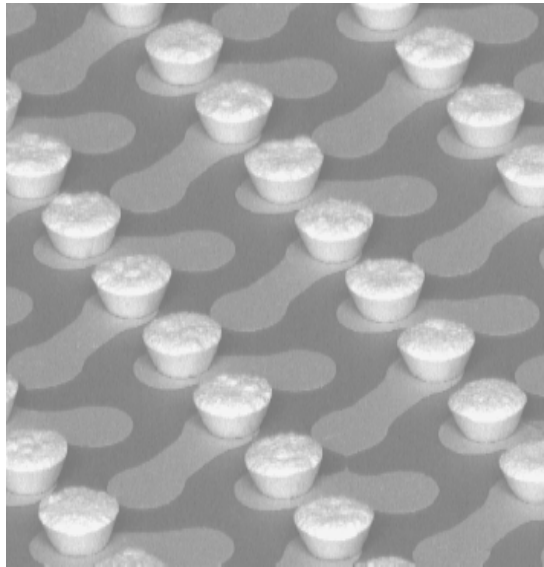
- Low contact resistance

Source: P.M. Martin et al., PNNL



JPL

Thermoelectric Module II



TE Materials

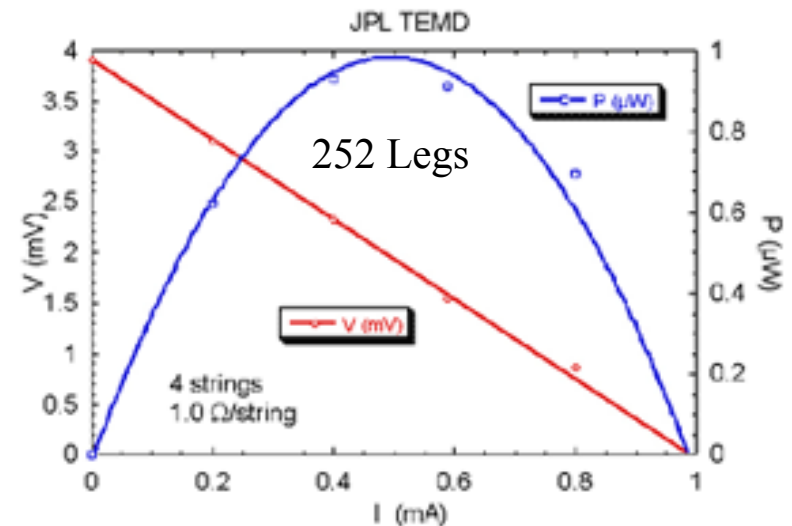
- Thickness ~ 100 's μm
- Small ΔT – Limited TE Thickness
- Electrical Conductivity

Substrate

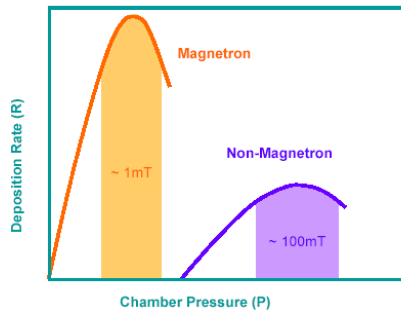
- Good thermal conductivity
- Electrically insulating

Interconnect Technology

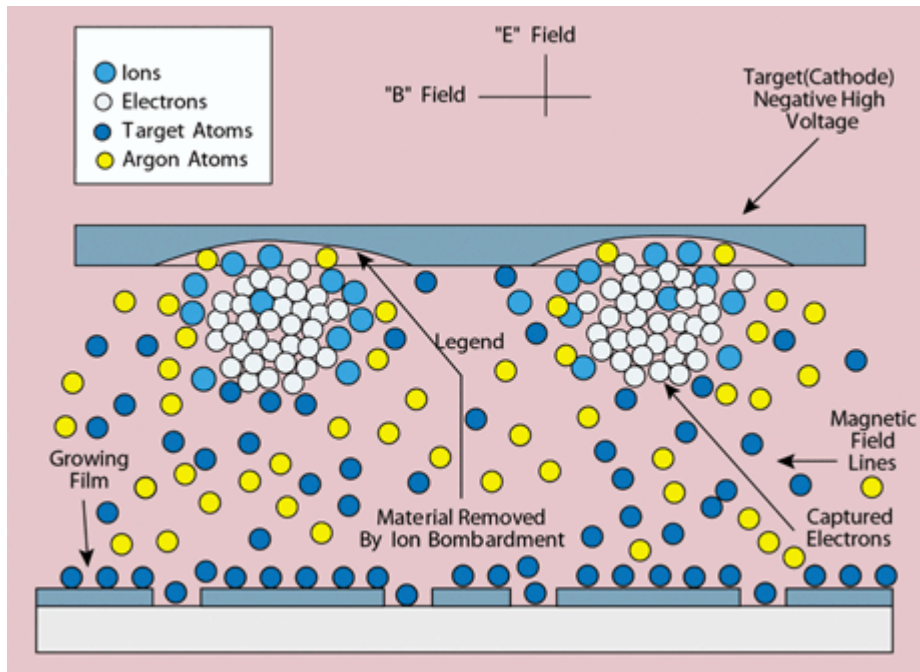
- Low contact resistance



Magnetron Sputtering



- Lower Deposition Pressure
- Reduces Contamination
- Increases Deposition Rate
- Complex Alloys



3 RF
Guns
600 W

Sample
Rotation Heater
RT – 800 °C



Magnetron Sputtering

Deposition Rate

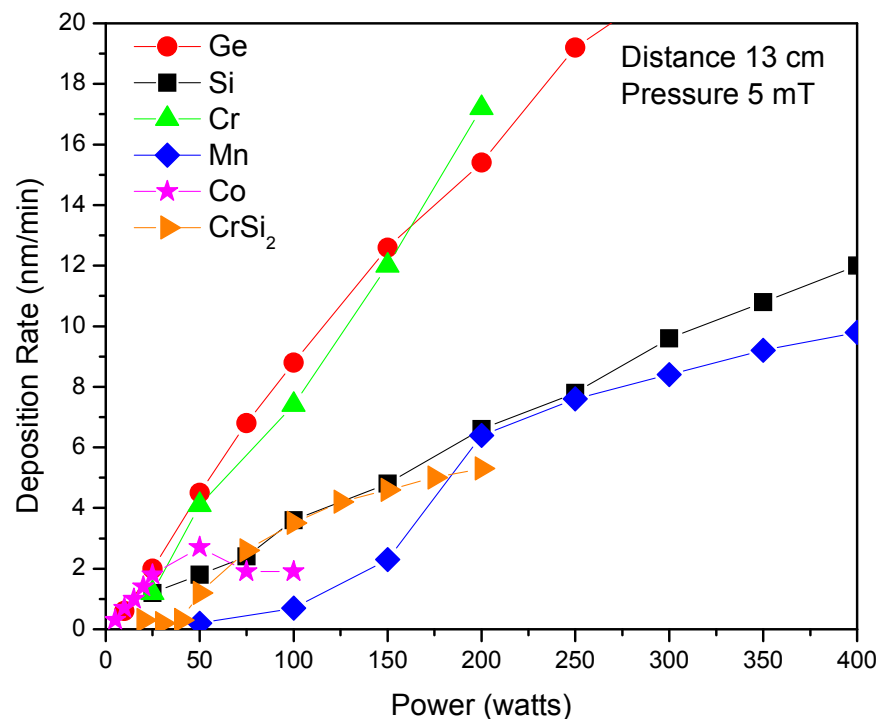
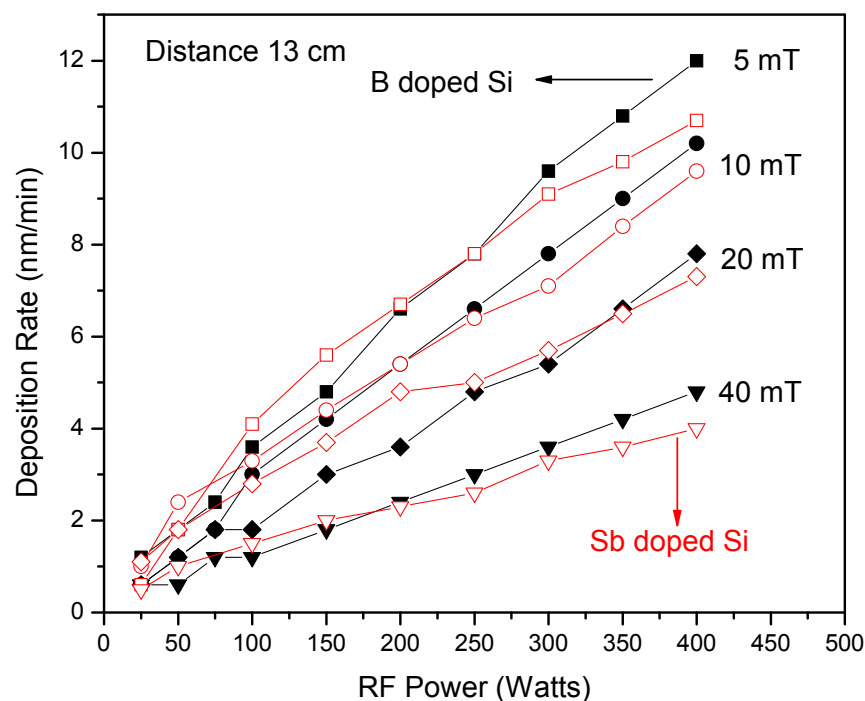
- Distance – $1/(d)^{1/2}$
- Material
- Pressure

Targets

- Commercial Sources
- Single Crystal – B-Si, Sb-Si

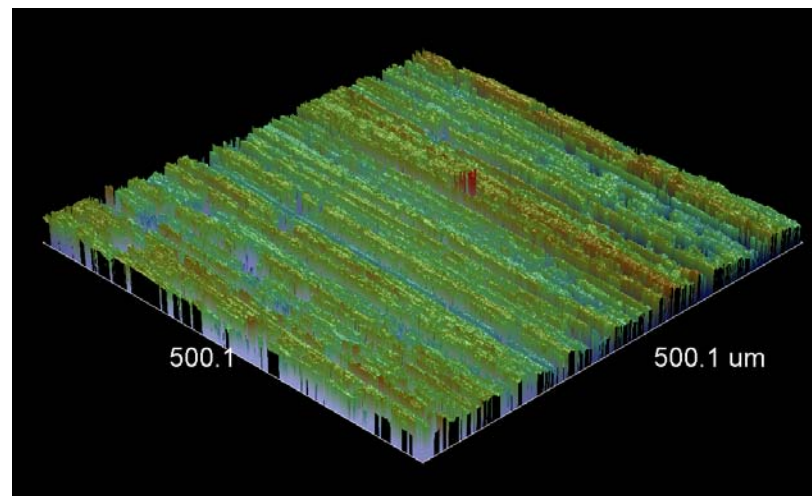
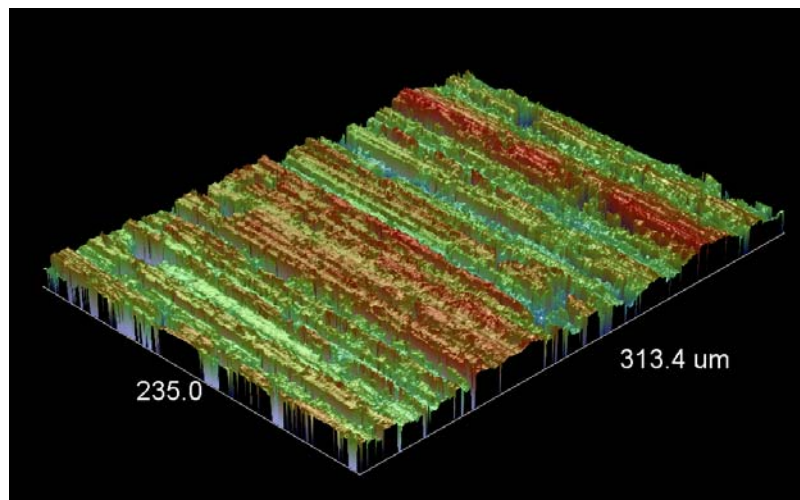
Co-Sputtering

Adjust Deposition Rate
To Control Composition



Y_2O_3 Stabilized ZrO_2 Substrate

Surface Roughness 500 nm



Deposit film retains the substrate surface roughness

YSZ ~ 2.5 W/mK Thermal Conductivity
 11×10^{-6} /K Thermal Expansion

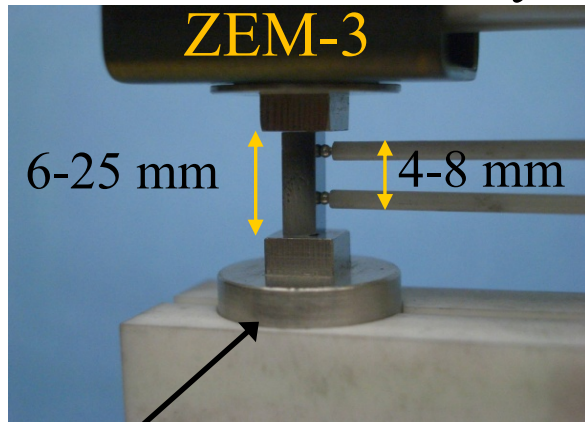
Total Film Stress = $\sigma_{\text{TE}} + \sigma_{\text{process}} + \sigma_{\text{external}}$

Film Failure is a Concern-Thickness $> 1 \mu\text{m}$



TE Property Measurements

Seebeck/Resistivity



ΔT 0-50 °C/Furnace RT-1000 °C

Hall Measurement

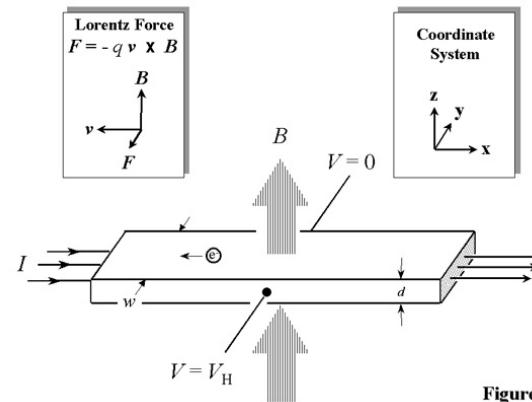
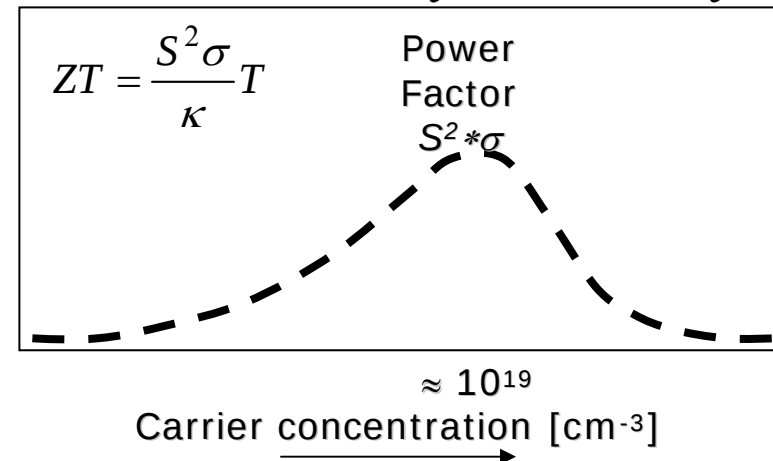


Figure 1

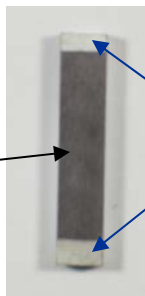
Capability-RT-527 °C/Field-2 Tesla

Carrier Mobility & Density



Silicide Film

Ag Electrodes



Silicon Films

Si, B-Si, Sb-Si & Si/Ge Films

- Amorphous Films
- Deposition Temperatures – up to 750 °C
- High Resistivity – $10^9 \Omega\text{-m}$
- Anneal – 800 °C - Resistivity – $10^6 \Omega\text{-m}$

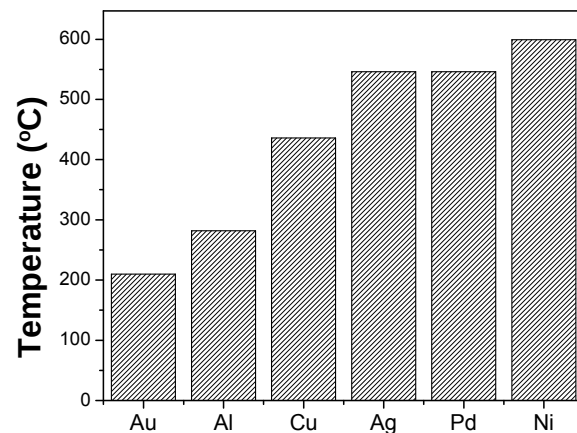
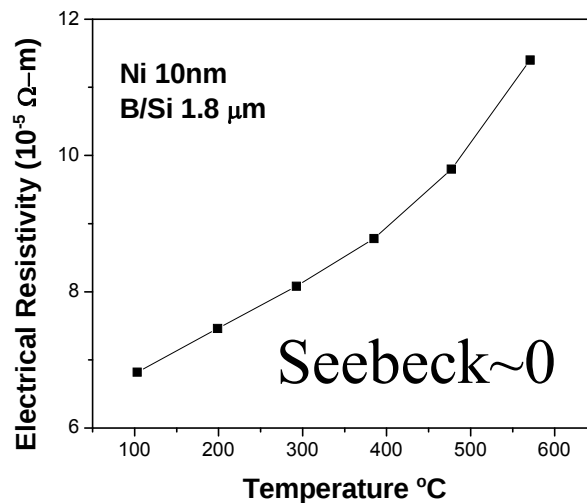
Metal Induced Crystallization

Silicon Film



Pre-layer
Ni,Cr

Lower Crystallization
Temperature



Knaepen et al., Thin Solid Films, 516, 4946, 2008

Mn-Si

Deposition Temperature – 450 °C

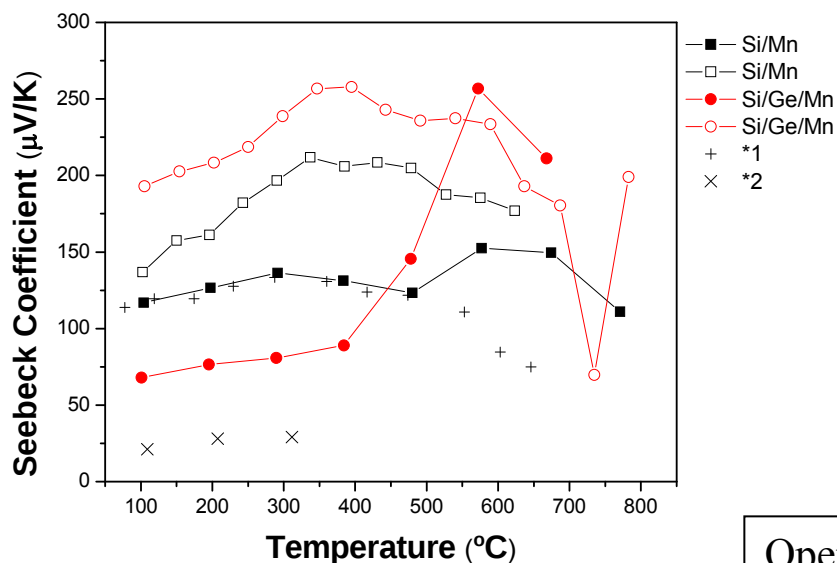
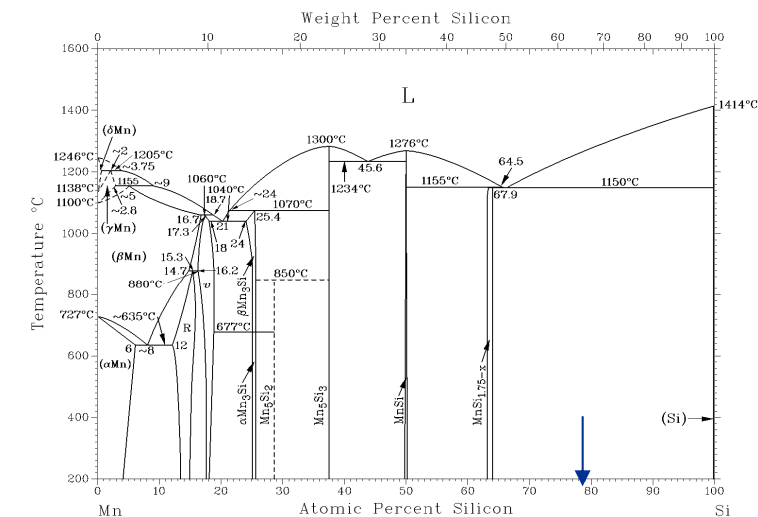
XRD – Crystalline Phases – $\text{Mn}_{15}\text{Si}_{26}$

Film Thickness – 2.4 μm & 2.6 μm

800 °C Anneal



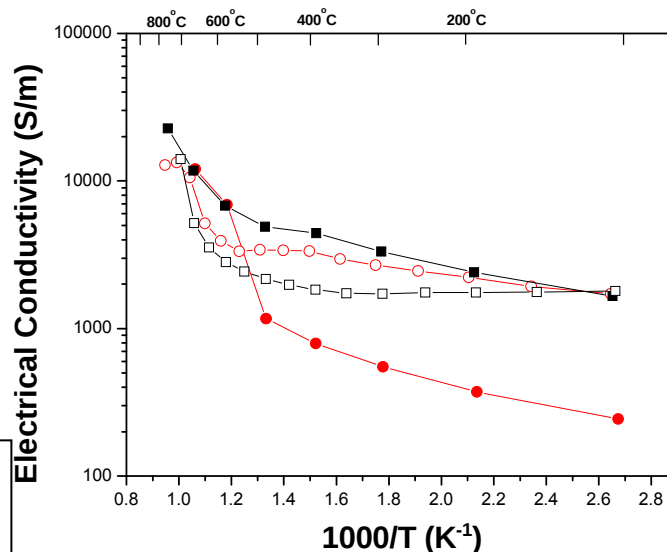
No Adhesion
Failures – Tape
Test



*1 Hou et al., Phy. Stat. Sol. A, 204, 3429, 2007

*2 Hou et al., Appl. Phys. A, 80, 1807, 2005

Open Symbols
2nd Measurement



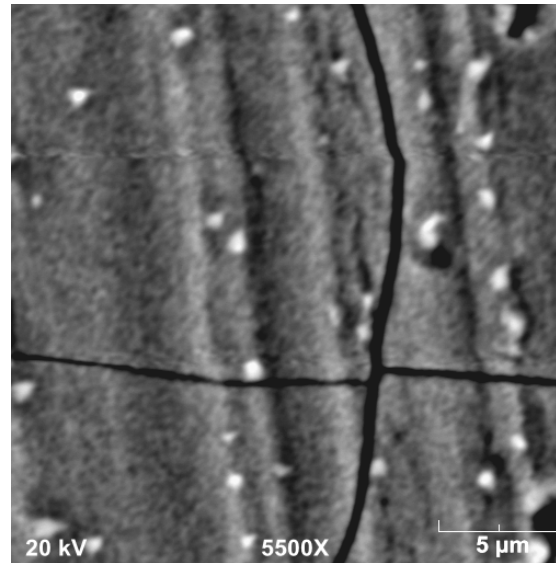
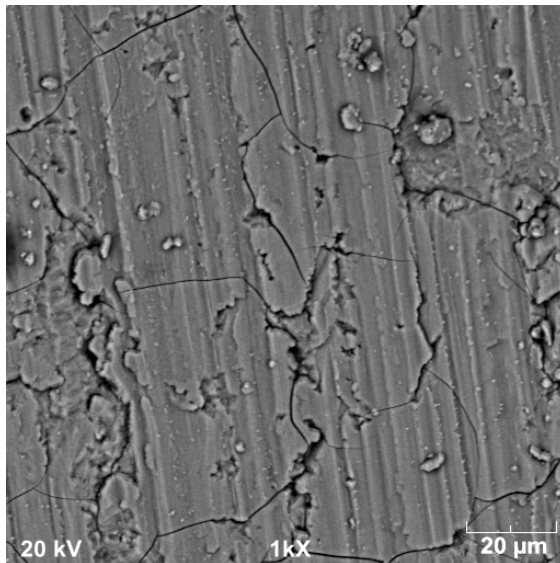
Mn-Si

Hall Measurements

Si/Mn 78/22 %	ρ ($10^{-5} \Omega\text{-m}$)	Carrier ($1/\text{cm}^3$)	Mobility (cm^2/VS)
As Deposit	14.1	2×10^{21}	0.25
600°C-4 hrs	31	5×10^{20}	0.5
800°C-2 hrs	242	2×10^{20}	0.1

B-Si/Ge/Mn 61/37/2 %	ρ ($10^{-5} \Omega\text{-m}$)	Carrier ($1/\text{cm}^3$)	Mobility (cm^2/VS)
As Deposit	280.1	6×10^{19}	0.3
600°C-4 hrs	40.8	6×10^{19}	2.6
800°C-2 hrs	50	2×10^{19}	7.3

Microstructure



- Film Cracking
- No observable Grain Structure
- Defects due to Substrate

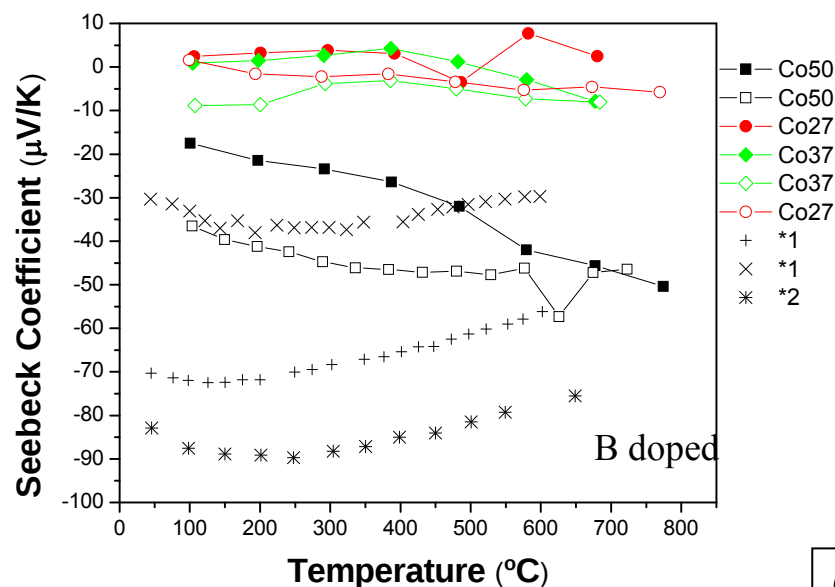
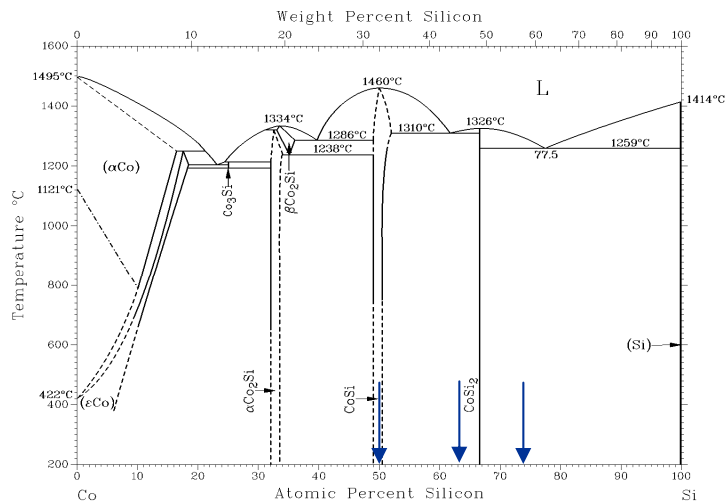
Co-Si

Deposition Temperature – 450 °C & 250 °C
Film Thickness – 1.5 μm & 1.8 μm

800 °C Anneal

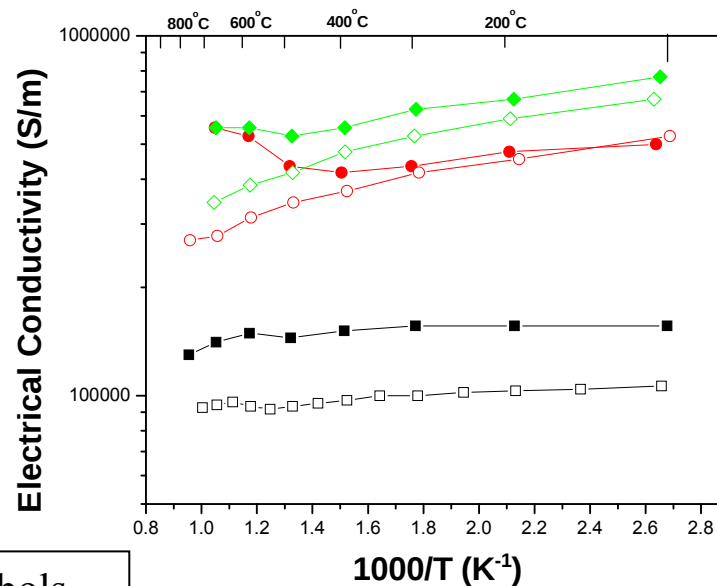


No Adhesion
Failures – Tape
Test



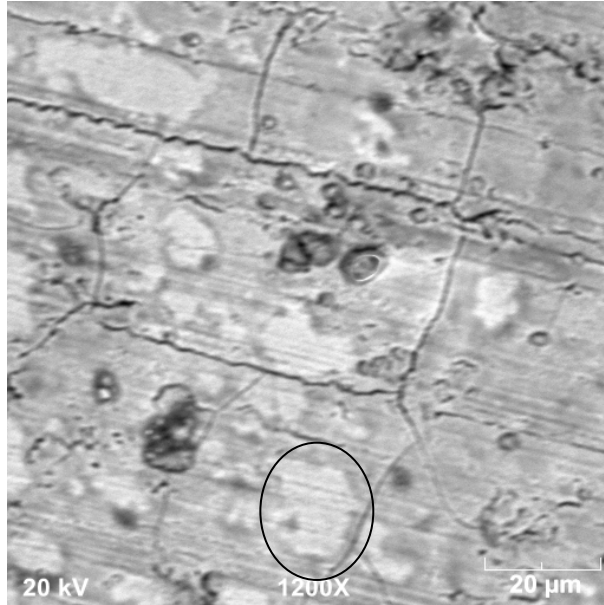
*1 Adachi et al., Mat. Sci. Forum, 426-432, 3445, 2003

*3 Ren et al., J. Alloys & Comp., 392, 50, 2005



Open Symbols
2nd Measurement

Hall Measurement



- Film Cracking
- No observable Grain Structure
- Defects due to Substrate
- Segregation?

Si/Co 50/50 %	ρ ($10^{-5} \Omega\text{-m}$)	Carrier ($1/\text{cm}^3$)	Mobility (cm^2/VS)
As Deposit	0.8	5×10^{20}	11
600°C-4 hrs	0.8	5×10^{20}	14
800°C-2 hrs	1.9	7×10^{20}	4

Sb-Si/Co 73/27 %	ρ ($10^{-5} \Omega\text{-m}$)	Carrier ($1/\text{cm}^3$)	Mobility (cm^2/VS)
As Deposit	0.3	1×10^{22}	2.5
600°C-4 hrs	0.1	1.5×10^{22}	5
800°C-2 hrs	0.1	1×10^{22}	3

Sb-Si/Co 62/37 %	ρ ($10^{-5} \Omega\text{-m}$)	Carrier ($1/\text{cm}^3$)	Mobility (cm^2/VS)
As Deposit	0.4	5×10^{21}	1.5
600°C-4 hrs	0.9	2×10^{22}	5
800°C-2 hrs	0.2	3×10^{22}	2

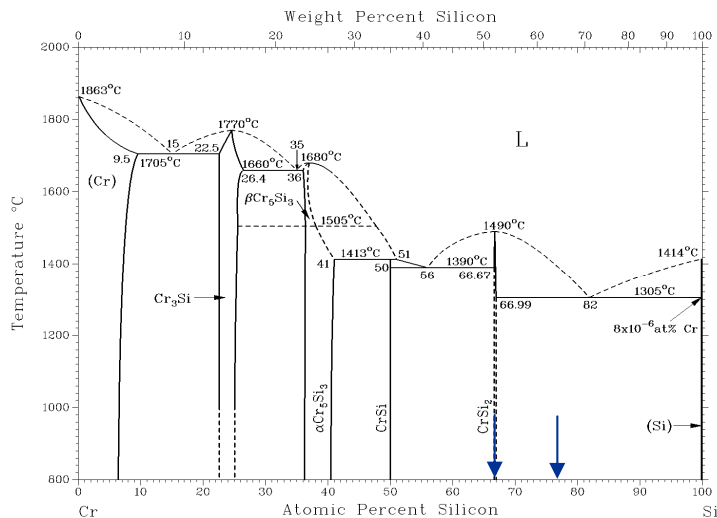
Cr-Si

Deposition Temperature – 450 °C

XRD – Target CrSi_2 - Poorly Crystallized

Co-sputtered – Crystalline CrSi_2

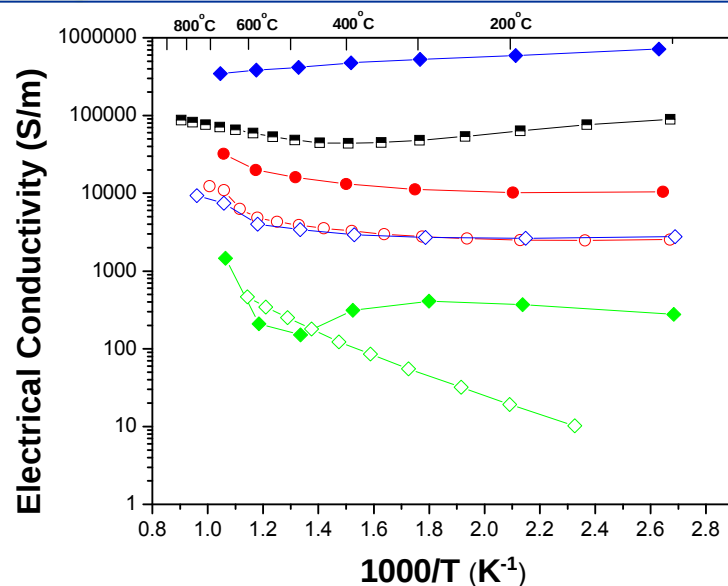
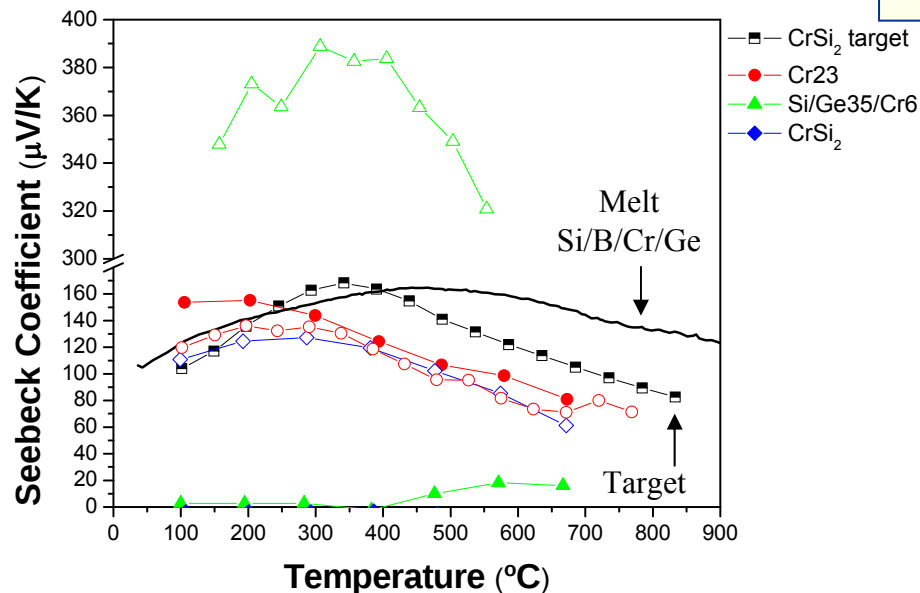
Film Thickness – 2.3 μm & 2.5 μm



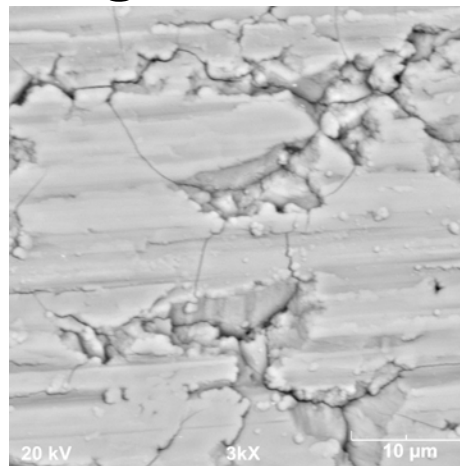
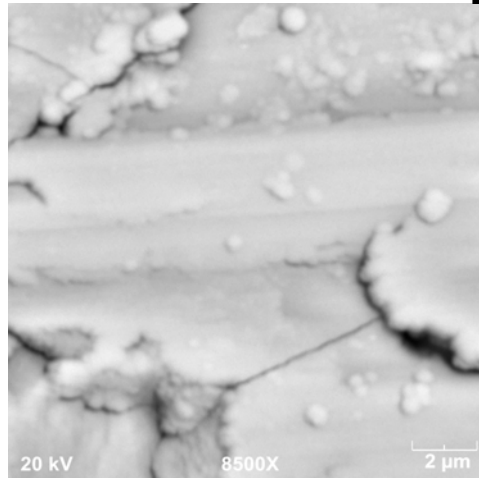
800 °C Anneal



No Adhesion
Failures – Tape
Test



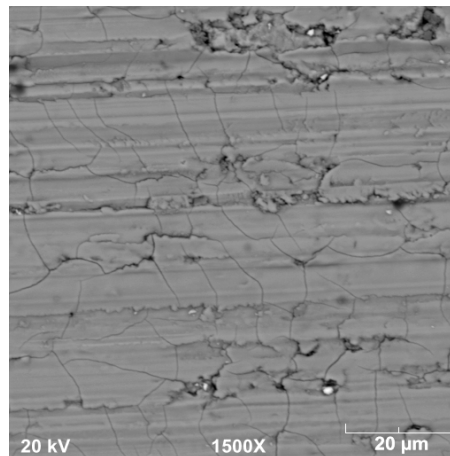
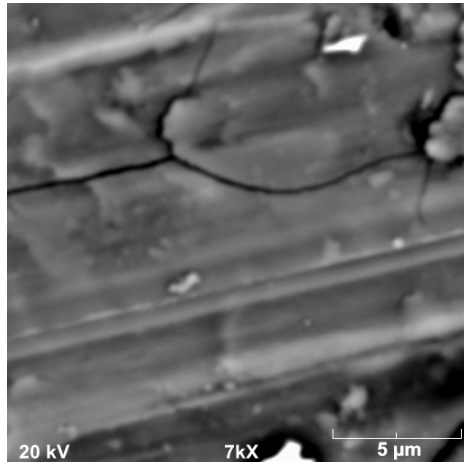
CrSi₂ Target



Hall Measurement

CrSi ₂	ρ ($10^{-5} \Omega\text{-m}$)	Carrier ($1/\text{cm}^3$)	Mobility (cm^2/VS)
As Deposit	9.2	1.5×10^{20}	1.4
600°C-4 hrs	37.3	1.7×10^{20}	1
800°C-2 hrs	775.4	6×10^{17}	10

Co-Sputtered



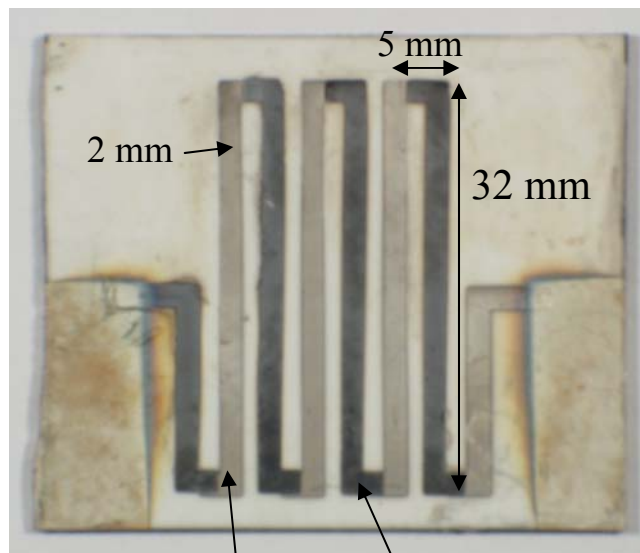
- Film Cracking
- No observable Grain Structure
- Defects due to Substrate

Hall Measurement

B-Si/Ge/Cr 59/35/6 %	ρ ($10^{-5} \Omega\text{-m}$)	Carrier ($1/\text{cm}^3$)	Mobility (cm^2/VS)
As Deposit	428	1.5×10^{21}	0.1
600°C-4 hrs	269×10^5	2×10^{17}	1
800°C-2 hrs	219	3.2×10^{18}	8.8

Simple Module

Shadow Mask



Si\Mn\Ge

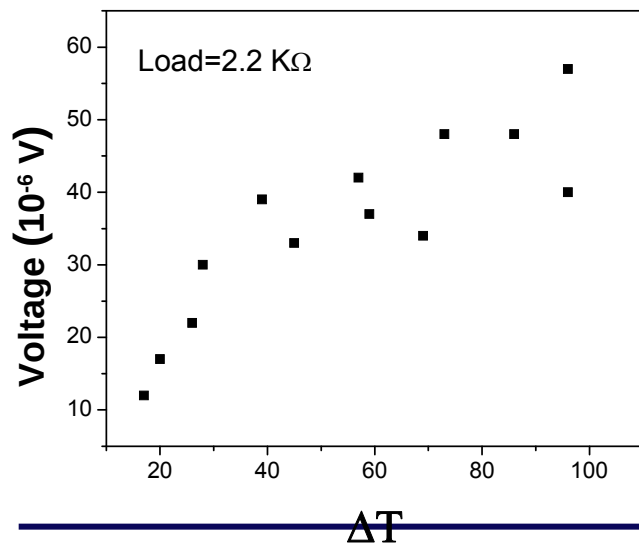
Large internal Ω

Legs $\sim 40 \text{ M}\Omega$

650 °C Anneal

N-leg Pt

Si\Mn\Ge



Power $\leq 10^{-12}$ watts



Summary

- Low resistive Si & Si/Ge difficult to achieve by magnetron sputtering.
- Low resistive films achieved with silicides of Co, Mn and Cr.
- Good Seebeck coefficients achieved for Mn & Cr silicides with P-type behavior.
- Moderate Seebeck coefficients achieved for Co silicides with N-type behavior.
- Silicide films exhibit low carrier mobility.
- Extensive film cracking is a problem.